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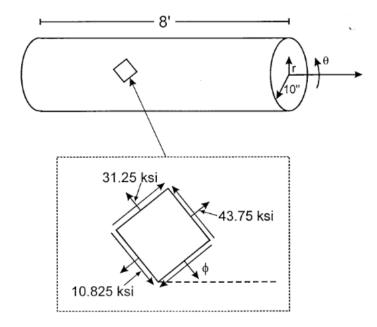
MSE 113 Mechanical Behavior of Materials

Midterm Exam #1

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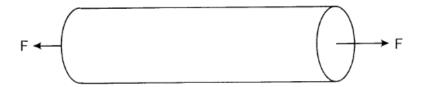
Problem	Total	Score
1	35	
2	30	
3	35	

Problem 1 35 points



Consider a thin walled, cylindrical tank made of 1080 steel under internal pressure with inside radius, r = 10 in, wall thickness, t = 0.05 in, and a length of 8 feet. Given the stresses acting on an element in the θ -z plane of the cylinder wall oriented at an angle ϕ from the coordinate axes,

- a) Find the angle φ that this element is oriented relative to the coordinate axes.
- b) Find $\sigma_{\theta\theta}$, σ_{zz} , and σ_{rr} .
- c) Find the internal pressure, P, in the cylinder.
- d) Find the axial force, F, that needs to be applied to the cylinder to reach the onset of yielding given that 1080 steel has a tensile yield strength of 85 ksi and ultimate tensile strength of 140 ksi. (Use Tresca criteria.) Hint, this is a 3D problem.



Solution

а

For the following equations consider: $\sigma_x = 43.75$, $\sigma_y = 31.25$, $\tau_{xy} = 10.825$

Center:
$$\frac{\sigma_x + \sigma_y}{2} = 37.5 \text{ ksi}$$

Radius:
$$\sqrt{\left(\frac{\sigma_x + \sigma_y}{2}\right)^2 + \tau_{xy}^2} = 12.5 \text{ ksi}$$

$$\tan(2\psi) = \frac{2\tau_{xy}}{\sigma_x - \sigma_y} = 1.732$$

$$\rightarrow \psi = 30^{\circ}$$

b.

$$\sigma_{max} = \sigma_{\theta\theta} = 37.5 + 12.5 = 50 \ ksi$$

$$\sigma_{min} = \sigma_{zz} = 37.5 - 12.5 = 25 \, ksi$$

$$\sigma_{rr} \approx 0$$

C

$$\sigma_{\theta\theta} = \frac{pr}{t}$$

$$p = \frac{\sigma_{\theta\theta}t}{r} = \frac{50,000 \ psi * .05"}{10"} = 250 \ psi$$

or

$$\sigma_{zz} = \frac{pr}{2t}$$

$$p = \frac{2\sigma_{\theta\theta}t}{r} = \frac{2 * 25,000 \ psi * .05''}{10''} = 250 \ psi$$

d.

Use superposition: Tension + Pressure

$$\tau_{max} = k = \frac{85 + 0}{2} = 42.5 \text{ ksi}$$

$$\rightarrow \sigma_{tensile\ yield} = 2 * \tau_{max} = 85\ ksi$$

For yielding:
$$\sigma_{zz} = 85 \ ksi = \sigma_{zz}^{tension} + \sigma_{zz}^{pressure}$$

$$\rightarrow \sigma_{zz}^{tension} = 85 - 25 = 60 \, ksi$$

$$\sigma_{zz}^{tension} = \frac{F}{Area} = \frac{F}{2\pi rt}$$

$$\rightarrow$$
 $F = \sigma_{zz}^{tension} * 2\pi rt = 60 \ ksi * 2 * \pi * 10" * .05" = 188.5 \ kips$

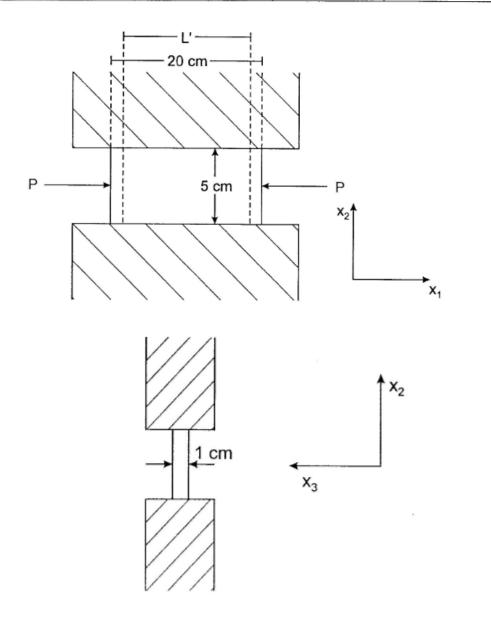
Problem 2 30 points

Suppose you have a long thin plane made of 6061-T4 aluminum between two rigid (non-deforming) walls acted on by a compressive force P of 10,000 N in the x_1 direction. The plate has original dimensions: L = 20 cm, b = 5 cm, and t = 1 cm.

Find σ_{22} , $\varepsilon_{33},$ and the new length L'.

Properties of 6061-T4 Aluminum

Young's			Tensile Yield	Ultimate Tensile
Modulus	Shear Modulus	Poisson's Ratio	Strength	Strength
(GPa)	(GPa)		(MPa)	(MPa)
70	26.3	1/3	145	240



$$\overline{\sigma_{11}} = \frac{-P}{bt} \quad \text{negative because it is a compressive force} \\
\sigma_{11} = \frac{-10,000 \, N}{0.05 \, m * 0.01 \, m}$$

This will produce a stress in the 2 direction, σ_{22} , because of the constraint, $\varepsilon_{22} = 0$. There won't be a stress in the 3 direction, $\sigma_{33} = 0$, because there is no constraint. Instead the material will deform in this direction, ε_{33} .

Note
$$\sigma_{33} = \sigma_{12} = \sigma_{13} = \sigma_{23} = 0$$

Setup strain equations:

$$\varepsilon_{11} = \frac{1}{E}(\sigma_{11} - v\sigma_{22}) \qquad \varepsilon_{22} = \frac{1}{E}(\sigma_{22} - v\sigma_{11}) \qquad \varepsilon_{33} = \frac{1}{E}(-v(\sigma_{11} + \sigma_{22}))$$

$$\varepsilon_{12} = \varepsilon_{13} = \varepsilon_{12} = 0$$

From
$$\varepsilon_{22} = 0 \rightarrow \sigma_{22} = v\sigma_{11} = \frac{1}{3} * -20 MPa = -6.67 MPa$$

Then,

$$\varepsilon_{33} = \frac{1}{E} \left(-v(\sigma_{11} + \sigma_{22}) \right)$$

$$\varepsilon_{33} = \frac{1}{70,000 \, MPa} \left(-\frac{1}{3} (-20 \, MPa - 6.67 \, MPa) \right)$$

$$\varepsilon_{33} = \mathbf{1}.\, \mathbf{27} * \mathbf{10}^{-4}$$

Finally,

$$\varepsilon_{11} = \frac{1}{E} (\sigma_{11} - \upsilon \sigma_{22})$$

$$\varepsilon_{11} = \frac{1}{70,000 \, MPa} \left(-20 \, MPa + \frac{6.67 \, MPa}{3} \right) = -2.54 * 10^{-4}$$

Where,
$$\varepsilon_{11} = \frac{\Delta L}{L}$$

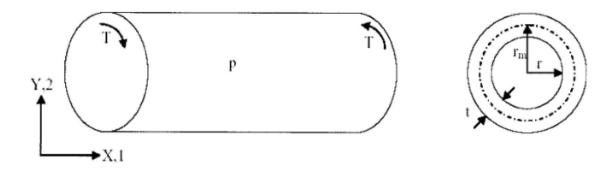
 $\rightarrow \Delta L = L * \varepsilon_{11} = 20 \ cm * -2.54 * 10^{-4} = -.005 \ cm$
 $L' = L + \Delta L = 20 \ cm - 0.005 \ cm = 19.995 \ cm$

Problem 3 35 points

The thin-walled cylinder, shown below, has an internal pressure of p = 1000 kPa, and is subjected to a twist of T = 10 MNm. The inner radius of the cylinder is 2 m with a wall thickness of 20 mm. The shear stress on a thin-walled tube can be approximated by:

$$\tau = \sigma_{\theta z} = \frac{T}{2 \pi r_m^2 t}$$

where T is the applied torsional moment, r_m is the radius to the median line, and t is the thickness of the cylinder. See below:



- a) Assuming that the ends have no effect on the stresses near the center of the cylinder,
 - i. Determine the principal stresses and show them on a sketch of a properly oriented element, i.e. rotated by the principal angle.
 - ii. Determine the maximum shear stress.
- b) Is there any evidence of plastic yielding in the cylinder if the yield strength of the material used is 180 MPa? (Consider only the θ -z plane, i.e. 2D)
- c) If the cylinder is punctured to leave a tiny pinhole in the wall thickness, check if yielding will occur at the edge of the hole using the Tresca and von Mises criteria.

$$\overline{r_m} = r + \frac{t}{2} = 2 + \frac{0.02}{2} m = 2.01 m$$

$$\tau = \sigma_{\theta z} = \frac{T}{2 \pi r_m^2 t} = \frac{10 * 10^6 \cdot N \cdot m}{2 \pi (2.01 m)^2 * 0.02 m} = 1.97 * 10^7 Pa = 19.7 MPa$$

$$\sigma_{\theta \theta} = \frac{pr}{t} = \frac{1 MPa * 2 m}{0.02 m} = 100 MPa$$

$$\sigma_{zz} = \frac{pr}{2t} = \frac{1 MPa * 2 m}{2 * 0.02 m} = 50 MPa$$

а

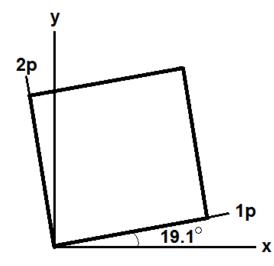
For the following equations,

$$\sigma_x = \sigma_{\theta\theta} = 100 MPa$$
 $\sigma_y = \sigma_{zz} = 50 MPa$
 $\tau_{xy} = \tau = 19.7 MPa$

$$\sigma_{1p} = \frac{\sigma_x + \sigma_y}{2} + \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2} = 106.8 MPa$$

$$\sigma_{2p} = \frac{\sigma_x + \sigma_y}{2} - \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2} = 43.2 MPa$$

$$\tan(2\theta_p) = \frac{2\tau_{xy}}{\sigma_x - \sigma_y} \rightarrow \theta_p = 19.1^\circ$$



$$\tau_{max} = \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2} = 31.8 MPa$$

b.

Tresca

$$k = \frac{\sigma_y}{2} = \frac{180 \text{ MPa}}{2} = 90 \text{ MPa} > \tau_{max} = 31.8 \text{ MPa}$$

Therefore, does **not** yield.

Von Mises

Consider the stress state in the θ -z plane, called the 1-2 plane here. Note that the stresses in the 3rd direction are zero since this is a 2D problem.

$$\sigma = \sqrt{\frac{1}{2}[(\sigma_{11} - \sigma_{22})^2 + (\sigma_{22} - \sigma_{33})^2 + (\sigma_{33} - \sigma_{11})^2] + 3(\sigma_{12}^2 + \sigma_{23}^2 + \sigma_{31}^2)}$$

$$\sigma = \sqrt{\frac{1}{2}[(100 - 50)^2 + (50 - 0)^2 + (0 - 100)^2] + 3(19.7^2 + 0^2 + 0^2)}$$

$$\sigma = 93.1 \, MPa < \sigma_v = 180 \, MPa$$

Therefore, does **not** yield.

c.

$$S_1 = \sigma_{1p} = 106.8 MPa$$
 $S_2 = \sigma_{2p} = 43.2 MPa$

$$\sigma_{A\theta\theta} = 3S_2 - S_1 = 22.7 \, MPa$$
 $\sigma_{B\theta\theta} = 3S_1 - S_2 = 277.3 \, MPa$

 $\sigma_{Arr} = \sigma_{Brr} = \sigma_{Ar\theta} = \sigma_{Br\theta} = 0$ Note that at the edge of a hole there is no material radially inward. If there is no material present, then there can't be can't be a stress inward. In order to have force balance, there can't be any radial stresses at the edge of a hole.

At point A

Tresca

$$\tau_{max} = \frac{\sigma_{A\theta\theta} - \sigma_{Arr}}{2} = 11.34 \text{ MPa} < 90 \text{ MPa} = k$$

Von Mises

$$\sigma = \sqrt{\frac{1}{2}[(22.7 - 0)^2 + (0 - 0)^2 + (0 - 22.7)^2] + 3(0^2 + 0^2 + 0^2)} = 11.34 < k$$

Yielding does not occur at point A.

At point B

Tresca

$$\tau_{max} = \frac{\sigma_{A\theta\theta} - \sigma_{Arr}}{2} = 138.7 \; MPa > 90 \; MPa = k$$

$$\sigma = \sqrt{\frac{1}{2}[(138.7 - 0)^2 + (0 - 0)^2 + (0 - 138.7)^2] + 3(0^2 + 0^2 + 0^2)} = 138.7 \, MPa > k$$
 Yielding occurs at Point B.

