

# ME 132, Spring 2003, Final Exam

Name:

# 1	# 2	# 3	# 4	# 5	# 6	# 7
10	20	15	15	15	10	15

# 8	# 9	# 10	# 11	# 12	# 13	TOTAL
15	20	15	15	15	15	180

- Do problems 1-11. Also do problem 12 or problem 13, but not both. Mark above which (of 12 and 13) one you want me to count in your exam.
- Any unmarked summing junctions are positively signed (+).

1. The input  $u$ , and output  $y$ , of a single-input, single-output system are related by

$$y^{[3]}(t) + 6y^{[2]}(t) + 2y^{[1]}(t) + 3y(t) = 2u^{[2]}(t) - 5u^{[1]}(t) - 5u(t)$$

(a) Find the transfer function from  $U$  to  $Y$

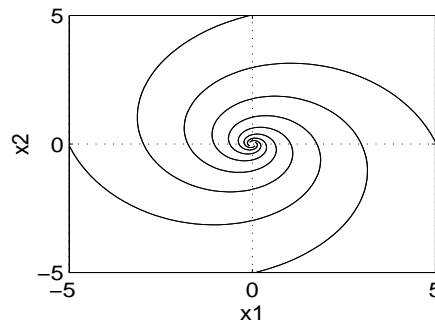
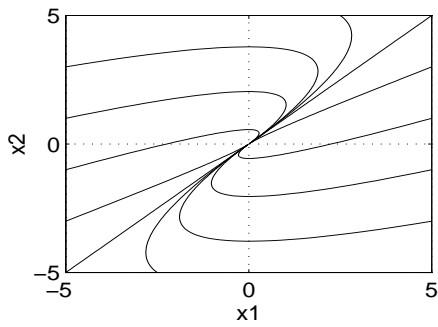
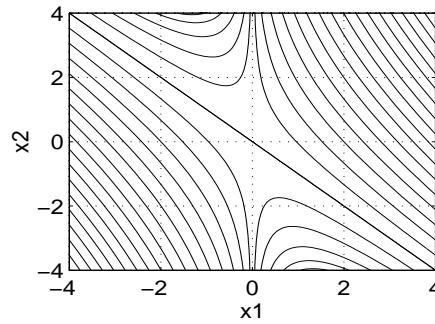
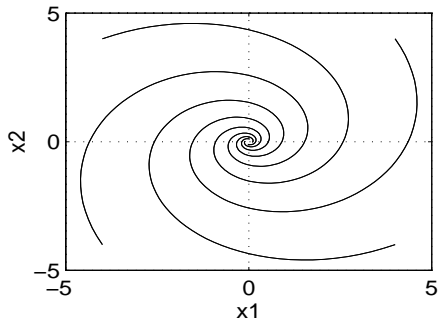
(b) Show that this is a stable system.

(c) If  $u(t) \equiv 2$  for all  $t \geq 0$ , what is the limiting value of  $y$ , namely  $\lim_{t \rightarrow \infty}$ ?

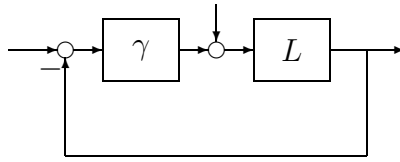
(d) Suppose the input is sinusoidal,  $u(t) = \sin(100t)$ . In the steady state, what is the approximate amplitude of the sinusoidal output  $y$ ?

2. Consider the 2-state system governed by the equation  $\dot{x}(t) = Ax(t)$ . Shown below are the phase-plane plots ( $x_1(t)$  vs.  $x_2(t)$ ) for 4 different cases. Match the plots with the  $A$  matrices, and correctly draw in arrows indicating the evolution in time. Put your answers on the enlarged graphs included in the solution packet.

$$A_1 = \begin{bmatrix} -3 & 2 \\ -1 & 0 \end{bmatrix}, \quad A_2 = \begin{bmatrix} -2 & 0 \\ 3 & 1 \end{bmatrix}, \quad A_3 = \begin{bmatrix} -1 & 3 \\ -3 & -1 \end{bmatrix}, \quad A_4 = \begin{bmatrix} 1 & 3 \\ -3 & 1 \end{bmatrix}$$



3. A feedback system is shown below.



Here  $\gamma$  is a constant real number. The open-loop transfer function is  $L(s)$ , given as

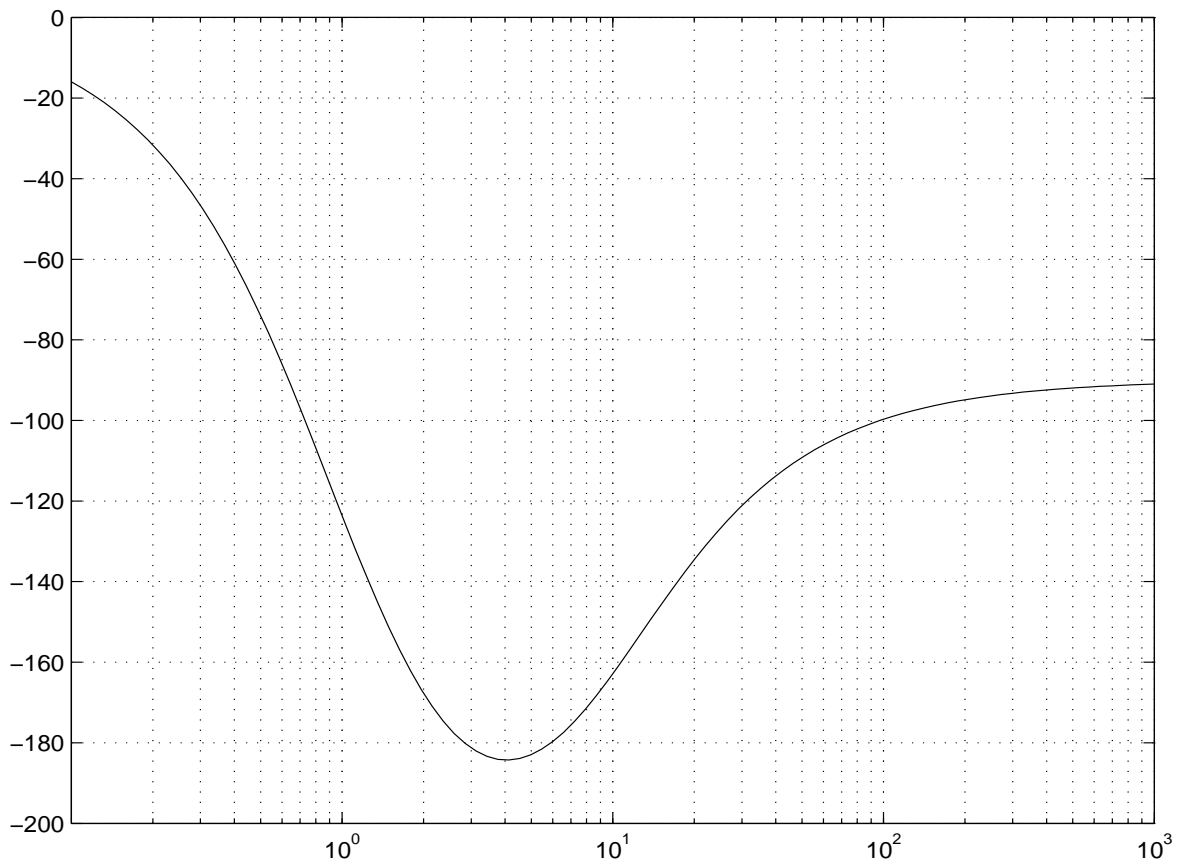
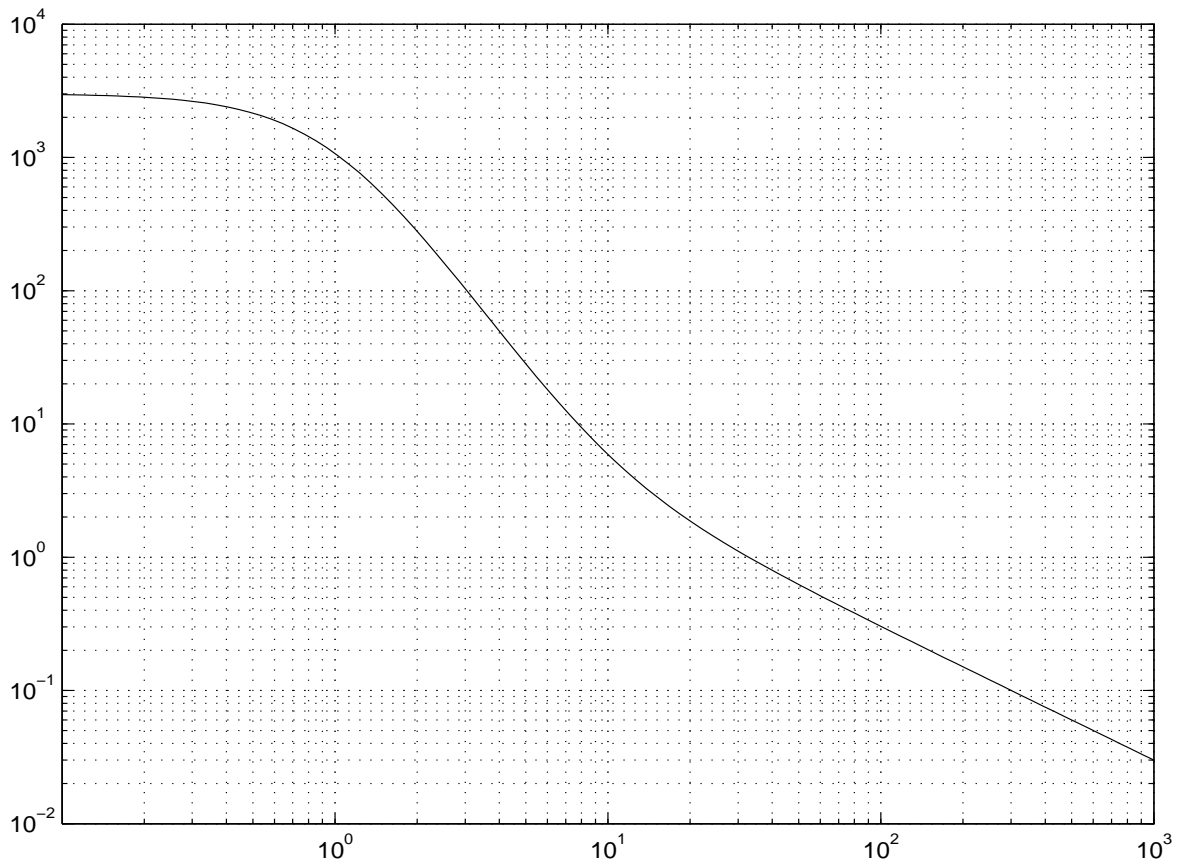
$$L(s) = \frac{30(s + 10)(s + 10)}{(s + 1)(s + 1)(s + 1)}$$

- (a) Determine the characteristic equation for the closed-loop system.
- (b) Using the 3rd order test for polynomials, completely determine the range of  $\gamma$  values that result in a stable system. **Here is a hint, provided simply to aid you in checking your answer:** I have already verified, for instance, that

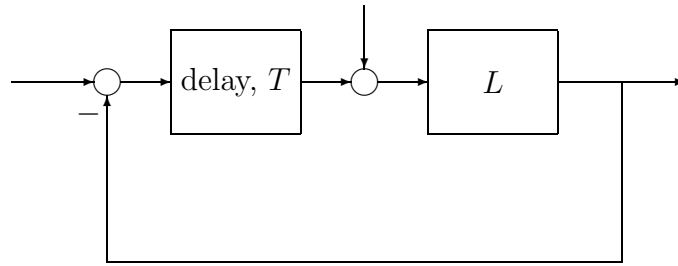
$\gamma$	Closed-loop is
-1	unstable
-0.1	unstable
-0.0002	stable
0.001	stable
0.01	unstable
0.1	stable
1	stable

4. In problem 3, you determined the overall possible ranges of  $\gamma$  for which the system shown below is stable. As you determined (and divulged in the **hint**), the stability region includes the point  $\gamma = 1$ .
- (a) Taking the nominal value of  $\gamma = 1$ , use the Bode plot of  $L$  (provided on the next page) to determine the gain margin of the system. **Show your work. Do not simply take the numbers from problem 3, and express them as a gain margin. Work out the answer independently, using the graph of  $L$ .**

- (b) Compare your answer in Problem 3 to the answer here. They are very closely related, but the gain-margin answer has “less” information. Explain.

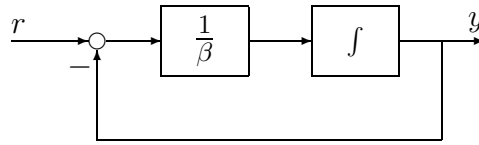


5. Take  $L$  from problem 3. There, you determined that for no time delay ( $T = 0$ ), the closed-loop system shown below is stable.



Use the Bode plot of  $L$  (provided on the previous page) to determine the time-delay margin of the system.

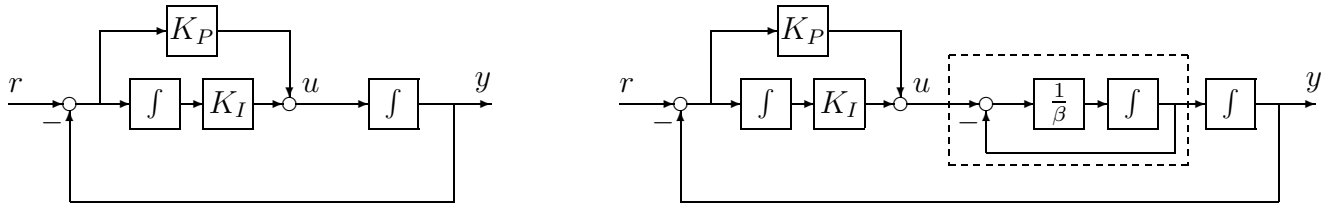
6. A block diagram is shown below.  $\beta$  is a real number.



- (a) What is the differential equation relating  $y$  and  $r$ ?
  
  
  
  
  
  
  
  
  
  
- (b) What is the transfer function from  $R$  to  $Y$ ?
  
  
  
  
  
  
  
  
  
  
- (c) Under what conditions (on  $\beta$ ) is the system stable?
  
  
  
  
  
  
  
  
  
  
- (d) If the system is stable, what is the time constant?
  
  
  
  
  
  
  
  
  
  
- (e) If the system is stable, what is the steady-state gain from  $r$  to  $y$ .



7. Shown below are two systems. The system on the left is the **nominal system**, while the system on the right represents a deviation from the nominal (the insertion of the dashed box) and is called the **perturbed system**.



Based on the values of  $K_P$  and  $K_I$ , and some analysis, you should have a general idea of how the nominal system behaves (eg., the effect of  $r$  on  $u$  and  $y$ ). Consider 3 different possibilities (listed below) regarding the relationship between the nominal and perturbed systems:

- (a) The perturbed system behaves pretty much the same as the nominal system.
- (b) The perturbed system behaves quite differently from the nominal system, but is still stable.
- (c) The perturbed system is unstable.

For each row in the table below, which description from above applies? Write **a**, **b**, or **c** in each box. Show work below.

$K_P$	$K_I$	$\beta$	Your Answer
2.8	4	0.02	
1.4	1	1	
14	100	0.2	
70	2500	0.02	

8. A process, with input  $v$  and output  $y$  is governed by

$$\dot{y}(t) - 2y(t) = v(t)$$

(a) What is the transfer function from  $V$  to  $Y$ ?

(b) Suppose  $y(0) = 1$ , and  $v(t) \equiv 0$  for all  $t \geq 0$ . What is the solution  $y(t)$  for  $t \geq 0$ . Is the process stable?

(c) Suppose that the input  $v$  is the sum of a control input  $u$  and a disturbance input  $d$ , so  $v(t) = u(t) + d(t)$ . Consider a PI control strategy,  $u(t) = K_P [r(t) - y(t)] + K_I z(t)$ ,  $\dot{z}(t) = r(t) - y(t)$ . Draw a block diagram of the closed-loop system using transfer function representations for the process and the controller. Include the external inputs  $r$  and  $d$ , and label the signals  $u$  and  $y$ .

(d) In the closed-loop, what are the transfer functions from  $R$  to  $Y$  and from  $D$  to  $Y$ .

(e) In the closed-loop, what are the transfer functions from  $R$  to  $U$  and from  $D$  to  $U$ .

(f) For what values of  $K_P$  and  $K_I$  is the closed-loop system stable?

(g) Choose  $K_P$  and  $K_I$  so that the closed-loop characteristic equation has roots with  $\xi = 0.707 (= \frac{1}{\sqrt{2}})$  and  $\omega_n = 2$ .

9. The pitching-axis of a tail-fin controlled missile is governed by the nonlinear state equations

$$\begin{aligned}\dot{\alpha}(t) &= f(\alpha(t)) \cos \alpha(t) + q(t) \\ \dot{q}(t) &= h(\alpha(t)) + Eu(t)\end{aligned}$$

Here, the states are  $x_1 := \alpha$ , the angle-of-attack, and  $x_2 := q$ , the angular velocity of the pitching axis. The input variable,  $u$ , is the deflection of the fin which is mounted at the tail of the missile.  $E$  is a physical constant, and  $E > 0$ .  $f$  and  $h$  are known, differentiable functions (from wind-tunnel data) of  $\alpha$ .

(a) Show that for any specific value of  $\bar{\alpha}$ , with  $|\bar{\alpha}| < \frac{\pi}{2}$ , there is a pair  $(\bar{q}, \bar{u})$  such that

$$\begin{bmatrix} \bar{\alpha} \\ \bar{q} \end{bmatrix}, \bar{u}$$

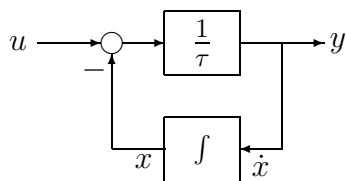
is an equilibrium point of the system (this represents a turn at a constant rate). Your answer should clearly show how  $\bar{q}$  and  $\bar{u}$  are functions of  $\bar{\alpha}$ , and will most likely involve the functions  $f$  and  $h$ .

(b) Calculate the Jacobian Linearization of the missile system about the equilibrium point. In other words, find a  $2 \times 2$  matrix  $A$ , and a  $2 \times 1$  matrix  $B$ , such that while they remain small, the deviation variables  $\delta_x(t) := x(t) - \bar{x}$ ,  $\delta_u(t) := u(t) - \bar{u}$  are approximately governed by

$$\dot{\delta}_x(t) = A\delta_x(t) + B\delta_u(t)$$

Your answers for  $A$  and  $B$  will be fairly symbolic, and may depend on the derivatives of the functions  $f$  and  $h$ . Be sure to indicate where the various terms are evaluated.

10. The block diagram below is often called an “approximate differentiator.” Note that nowhere in the block diagram is there a differentiating element.



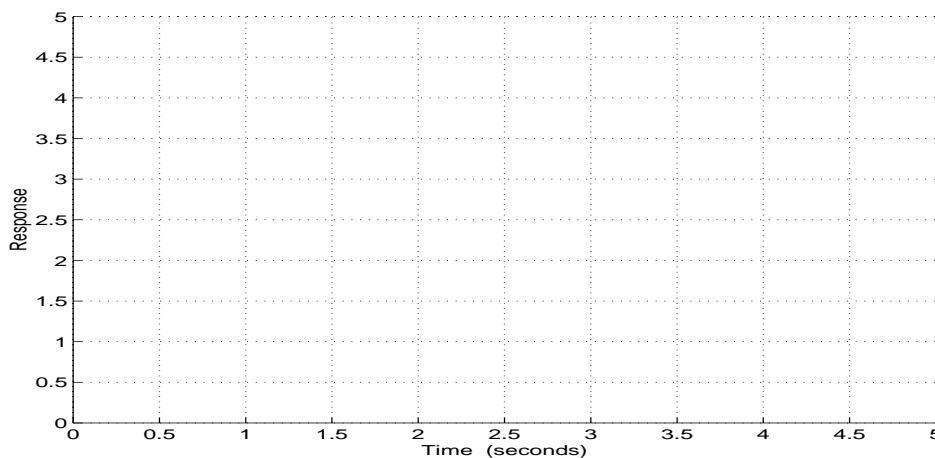
- (a) Based on the block diagram, write the differential equation relating  $x$ ,  $\dot{x}$  and  $u$ .

- (b) Write the equation expressing  $y$  in terms of  $x$  and  $u$

- (c) Show that the transfer function from  $U$  to  $Y$  is  $\frac{s}{\tau s + 1}$ .

(d) Suppose that the initial condition is  $x(0) = 0$ . Apply a step input at  $t = 0$ , so  $u(t) = \bar{u}$  for  $t > 0$  (here,  $\bar{u}$  is just some constant value). Compute the response  $x(t)$ , for  $t \geq 0$ .

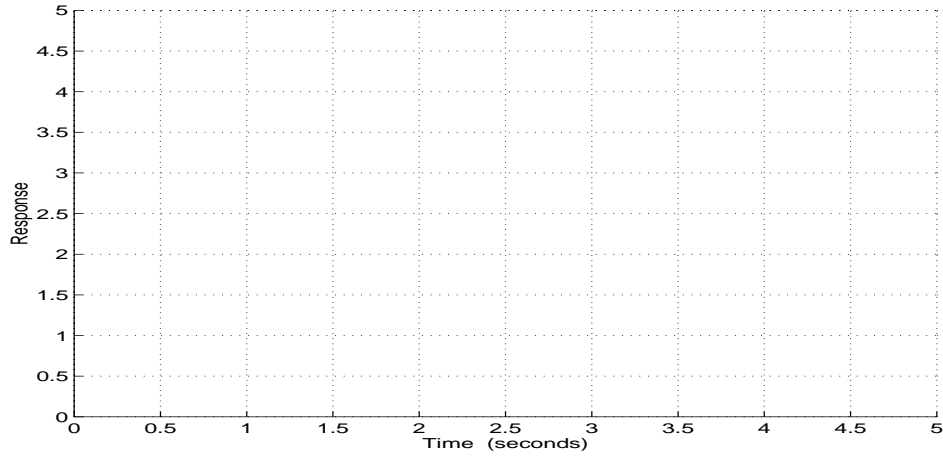
(e) With  $x(t)$  computed above, compute the output  $y(t)$ , and sketch below.



- (f) Suppose that the initial condition is  $x(0) = 0$ , let  $\tau = 0.2$ . Apply a ramp input (with slope 3)

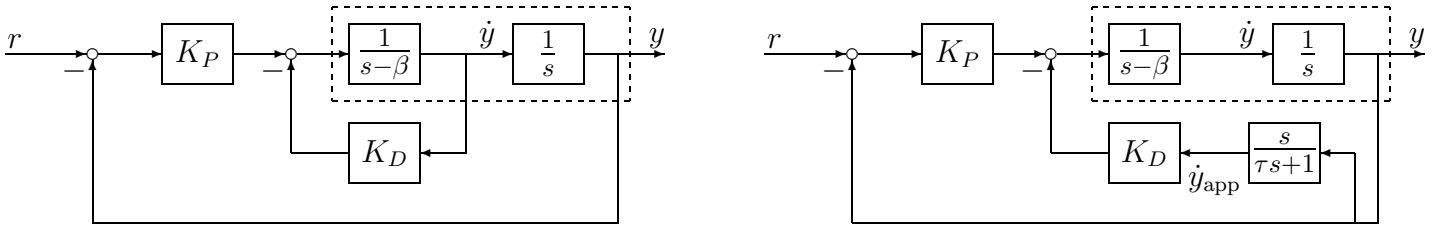
$$u(t) = 3t \text{ for } t \geq 0.$$

Compute the response  $y(t)$ , and plot. If you cannot derive the expression for  $y$ , guess what it should look like, and plot it below.

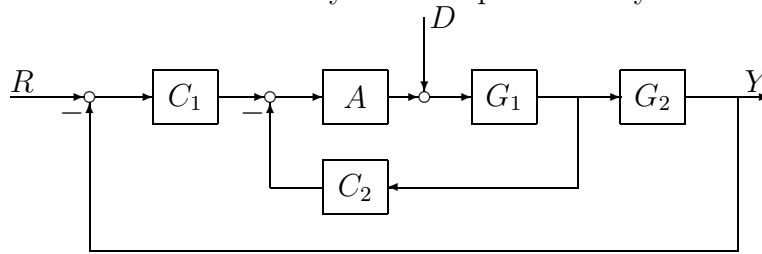




11. Block diagrams for two systems are shown below. Two of the blocks are just gains, ( $K_P$  and  $K_D$ ) and the other blocks are described by their transfer functions. The constant  $\beta$  is positive,  $\beta > 0$ . The system on the left is stable if and only if  $K_P > 0$  and  $K_D > \beta$  (no need to check this – it is correct). What are the conditions on  $K_P, K_D$  and  $\tau$ , such that the system on the right is stable? **Hint:** Note that  $\tau$  is the time-constant of the filter in the approximate differentiation used to obtain  $\dot{y}_{\text{app}}$  from  $y$ . The stability requirements will impose some relationship between its cutoff frequency  $\frac{1}{\tau}$  and the severity (eg., speed) of the unstable dynamics of the process, namely  $\beta$ .

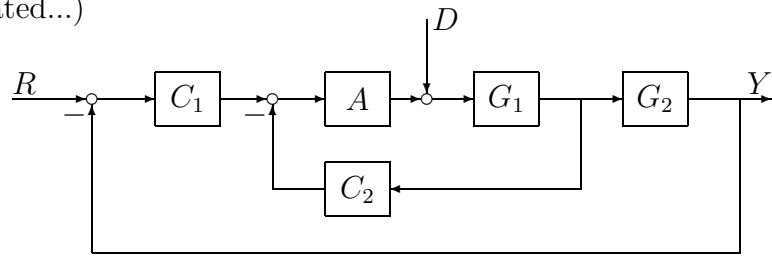


12. A block diagram is shown below. Each system is represented by its transfer function.



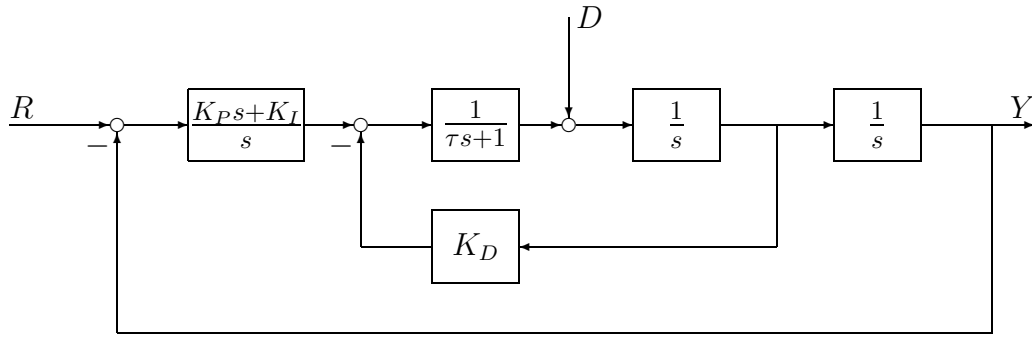
- (a) In terms of the transfer functions of the individual blocks, what is the transfer function from  $R$  to  $Y$ ?

(picture repeated...)



- (b) In terms of the transfer functions of the individual blocks, what is the transfer function from  $D$  to  $Y$ ?

13. A block diagram is shown below. Each system is represented by its transfer function.



- (a) In terms of  $K_P$ ,  $K_I$ ,  $K_D$ ,  $\tau$ , what is the transfer function from  $R$  to  $Y$  (hint: the denominator should be 4th order).

(b) In terms of  $K_P, K_I, K_D, \tau$ , what is the transfer function from  $D$  to  $Y$  (hint: same as above).

(c) In terms of  $K_P, K_I, K_D, \tau$ , what is the characteristic polynomial of the closed-loop system?

(d) In terms of  $K_P, K_I, K_D, \tau$ , what is the differential equation relating  $r$  and  $d$  to  $y$ ?