

Solutions to Spring 2004-2 Thermodynamics 105:

1) Solution:

1a): This is a steady flow problem. The enthalpy change in stream 1 should be equal to the enthalpy change in stream 2

$$(h_1(w) - h_2(w)) \times \text{mass flow rate of water} = \\ \text{Mass flow rate of air} \times (h_2(\text{air}) - h_1(\text{air})) = 29 \times 1 (T_2 - T_1) = 29 \times T_2$$

$$\text{Stream 1: } 29 \times 1 \times (T_{\text{mix}} - 0^\circ\text{C}) \\ \text{For air } \Delta H = 29 \times 1 \times 99.63 = 2889.27 \text{ kJoules} \\ \text{For steam } 2889.27 = 18 \times (h_1(w) - h_2(2)) \rightarrow h_2(w) = 2514.98 \text{ KJ/Kg} \\ \text{From steam table } \rightarrow \text{quality} = 0.929$$

1b) The entropy lost by stream 1 should equal the entropy gained by stream 2, an ideal heat engine, e. g. Carnot engine does not generate entropy

$$\text{Entropy change of stream 2 (air) is } \int (C_p/T) dT = 29 \times 1 \times \ln(373/273) = \\ 29 \times 0.312 = 9.05 \text{ kJ/K-min} = \text{entropy change of the air.}$$

Stream 1 still at 99.63C, some of the steam condenses, with entropy change: $\Delta S_{fg} = 6.04 \text{ kJ/(kg-K)}$
Thus: $9.05 = 18 \times 6.04 \text{ kJ/(kg-K)} \times (1-x)$
means that steam quality is $x = 0.916$

We now compute power by enthalpy balance: $18 \times \Delta H_{fg} \times (1-x)$ with $x = 0.916$
 $\Rightarrow 3382 \text{ kJ/min} = 56.4 \text{ kW}$ while air enthalpy change is same as part 1. 2889.27 kJ/min

Power is $3382 - 2889 \sim 482 \text{ kJ/min} \sim 8 \text{ kW}$

Note that slightly more steam condenses in Part 1b (8.3%) compared to Part 1a (7%).

2) Solution

A) Work = Heat * thermal efficiency
Thermal efficiency = $(1 - 1/8^{1.4-1}) = 56.47\%$
Work = $750 \text{ kJ} \times 0.5647 = 423.5 \text{ kJ}$

B) Heat = $750 \text{ kJ} \times (\text{mass of air with turbo-charger}) / \text{mass of air without turbo-charger}$
Work = Heat x thermal efficiency (same as it only depends on compression ratio)

$$\text{Mass of air with turbo-charger} / \text{mass of air without turbo-charger} \\ = (V_{\text{cylinder}}/v)_{\text{with turbo}} / (V_{\text{cylinder}}/v)_{\text{without turbo}} \\ = v_{\text{without turbo}} / v_{\text{with turbo}} \\ = (T/P)_{\text{without turbo}} / (T/P)_{\text{with turbo}}$$

$$\begin{aligned} & \text{(using isentropic relation } (T_2/T_1) = (P_2/P_1)^{(k-1)/k} \\ T_{\text{with turbo}} &= 300\text{K} * (120\text{kPa}/100\text{kPa})^{(1.4-1)/1.4} = 316\text{K} \\ &= (300\text{ K} / 100\text{kPa}) / (316\text{K} / 120\text{ kPa}) = 1.139 \end{aligned}$$

The amount of mass in the cylinder at BDC is increased by a factor of 1.139

$$\text{Work} = 750\text{kJ} \times 1.139 \times 0.5647 = 482.4\text{ kJ}$$

$$3) 1^{\text{st}} \text{ law } \rightarrow U_f = U_i \rightarrow m_1 u_1 + m_2 u_2 = M u_3 \rightarrow T_3 = (T_2 + T_1) / 2$$

Entropy change for partition 1 to final state

$$\Delta S_{3-1} = M/2 [C \ln (T_3/T_1)]$$

Entropy change for partition 2 to final state

$$\Delta S_{3-2} = M/2 [C \ln (T_3/T_2)]$$

$$\begin{aligned} \Delta S &= \Delta S_{3-1} + \Delta S_{3-2} = M/2 [C \ln (T_3/T_1) + C \ln (T_3/T_2)] \\ &= MC/2 \ln \{ T_3/T_1 * T_3/T_2 \} \\ &= MC \ln \{ T_3 / (T_1 * T_2)^{0.5} \} = MC \ln \{ (T_2 + T_1) / [2(T_1 * T_2)^{0.5}] \} \end{aligned}$$

$$4) \text{ At the inlet } p_{g1} = 0.8721 \text{ kPa} \rightarrow$$

$$W1 = 0.622 * \phi p_{g1} / (p - \phi p_{g1}) = 0.0040951 \text{ kgH}_2\text{/Kg dry air}$$

$$\text{At the exit, } p_{g2}(T=65^\circ\text{C}) = 25.03 \text{ kPa.}$$

$$\text{Now } \phi_2 = w_2 p_2 / (0.622 + w_2) / p_{g2} = 0.13065 \rightarrow 13.06\%$$

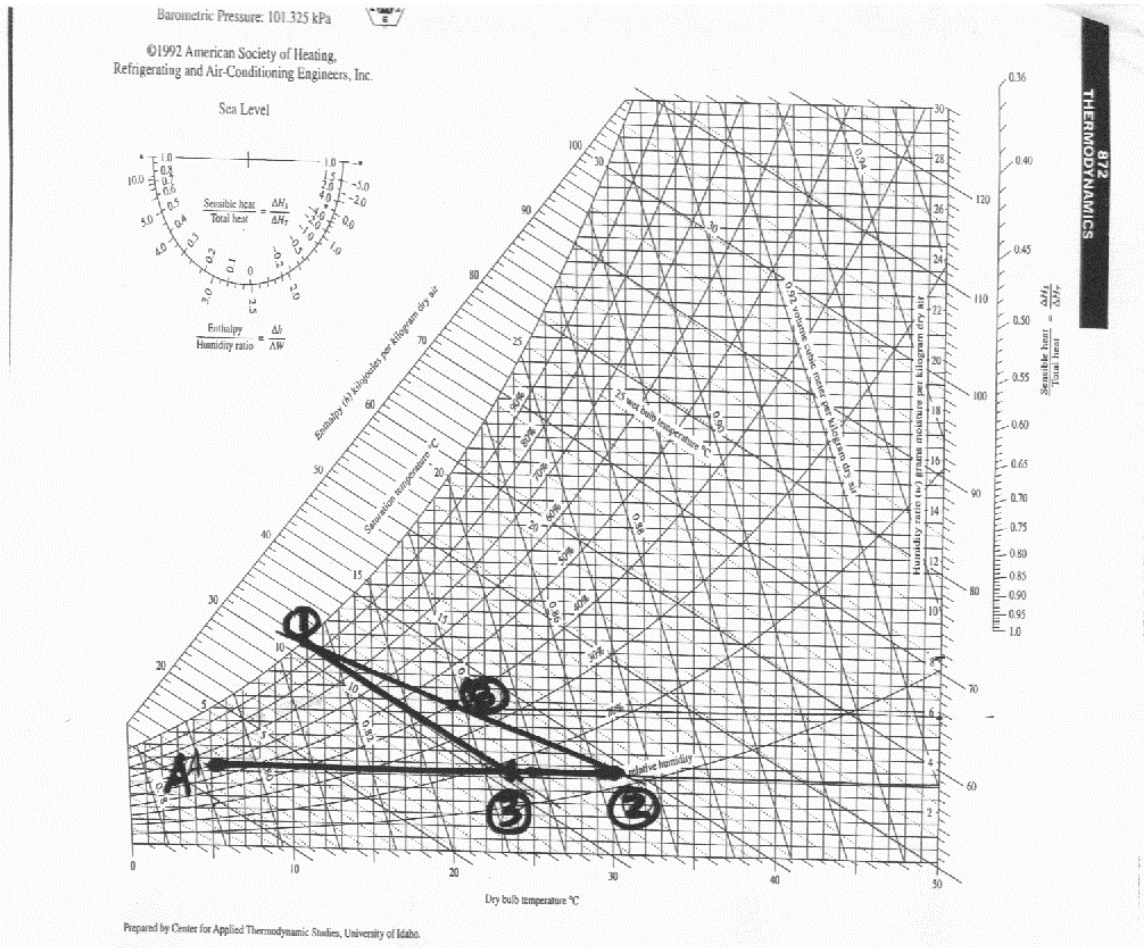
5) solution:

from the psychrometric chart state 2: $w_2 = 3.4 \times 10^{-3} \text{ kg/kg}$

by connecting the final point B to state 2 crossing the saturated state (state 1), we have $w_1 \sim 8 \times 10^{-3}$, $w_B = 5.5 \times 10^{-3} \text{ kg/kg}$,

a) the temperature at state 3 is obtained from state 1 (saturation point) following the constant enthalpy line until intersecting with the horizontal line (that connects state A and 2) $\rightarrow T_3 \sim 23.5^\circ\text{C}$

$$b) \text{ The mass flow rate ratio } m_{a1}/m_{a2} = (w_B - w_2) / (w_1 - w_B) \sim 0.96$$



6) By connecting a line with the two points in the psychrometric chart, we find the line intersects the saturation line; therefore condensation will occur during mixing.

7) From the R134-a tables:

State 3: saturated liquid $P=1.0164\text{MPa}$, $h_3=105.3\text{ kJ/kg}$,

State 3a: compressed liquid $\rightarrow h_{3a}=98.05\text{ kJ/kg}$

State 4a: saturated vapor $\rightarrow h_{4a}=235.31\text{ kJ/kg}$

State 4: constant enthalpy throttling $\rightarrow h_4=h_3$

Control volume analysis for the subcooler $\rightarrow h_1-h_{4a}=h_{3a}-h_3$

$h_1=265.76$; from superheated table $\rightarrow v_1 \sim 0.1627\text{ m}^3/\text{kg}$

2) mass flow rate of R134-a = $1.2\text{ m}^3/\text{min} / v_1 = 7.38\text{ kg/min} = 0.123\text{ kg/s}$

3) QL rate = $7.38\text{ kg/min} (235.31-98.05)\text{ KJ/kg} = 1012.97\text{ kJ/min} = 16.88\text{ kW}$

4) Work required by the compressor:

$s_1 = s_2 \sim 1.04\text{ kJ/kg-K}$

$p_2=1.0164\text{ Mpa} \rightarrow$ use the superheated table at 1.0MPa

$h_2 \sim 313.2\text{ kJ/kg}$

$W_{\text{compressor}} = 7.38\text{ kg/min} \times (313.6-265.76)\text{ kJ/kg} = 442.8\text{ kJ/min} \sim 7.38\text{ kW}$

$\text{COP} = \text{QL}/W_{\text{compressor}} = 16.88/7.38 = 2.29$

8) Solution:

For gas turbine cycle, $P_2=P_3$, and $P_4=P_1$, we have the following relation

$$(T_2/T_1)=(P_2/P_1)^{(k-1)/k}=(P_3/P_4)^{(k-1)/k}=(T_3/T_4).$$

When $T_2=T_4$, we have

$$T_1 T_3 = T_2^2$$

With given $T_3/T_1=3$ the above equation gives

$$3 T_1 \times T_1 = T_2^2 \rightarrow T_2/T_1 = (3)^{1/2}$$

Using the isentropic relation,

$$P_2/P_1 = (T_2/T_1)^{k/(k-1)} = 6.8$$